REPORT DOCUMENTATION PAGE

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For presentation at Direct Simulation Monte Carlo 2017; Santa Fe, NM, USA; 28 August 2017

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14. ABSTRACT

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VARIABLE WEIGHT FRACTIONAL COLLISIONS FOR MULTIPLE SPECIES MIXTURES

Robert Martin

IN-SPACE PROPULSION BRANCH, AIR FORCE RESEARCH LABORATORY, EDWARDS AIR FORCE BASE, CA USA

Direct Simulation Monte Carlo, 2017
Distribution Statement A: Approved for public release;
Distribution is Unlimited: PA #17517







OUTLINE



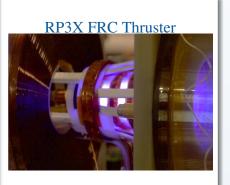
- BACKGROUND & REVIEW OF METHOD
- Multi-Species Test Cases
- 3 FUTURE WORK
- 4 Conclusion





Field-Reversed Configuration:

- Concept from Fusion Energy
 - Scaled Down for Propulsion





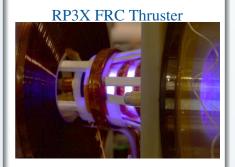


Field-Reversed Configuration:

- Concept from Fusion Energy
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- Electrodeless
 - Limits Erosion
 - Enables Flexible Fuels



Pancotti, et al. "Adaptive Electric Propulsion for ISRU Missions", 20th Adv. Space Prop., 11/2014

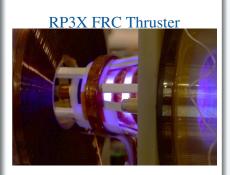






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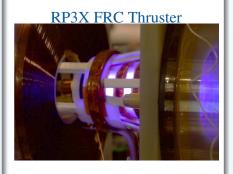






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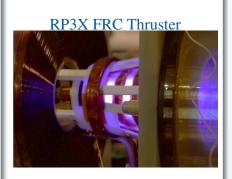




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Complex to Design



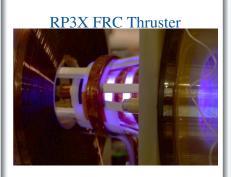




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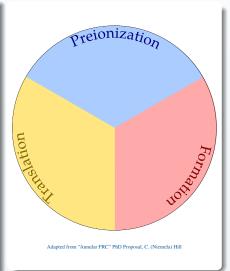
Complex to Design Significant Modeling Challenge







Dominant Physics Varies with Cycle:

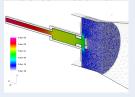


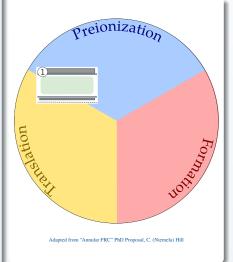




Dominant Physics Varies with Cycle:

Neutral FillRarefied Kinetic Flow



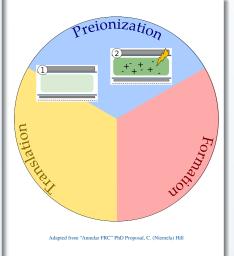






Dominant Physics Varies with Cycle:

- Neutral Fill
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- Preionization Chemistry
 - CR-Excitation/Ionization

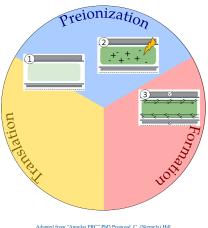






Dominant Physics Varies with Cycle:

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- Oriver Pulse
 - Ionization+Electromagnetics



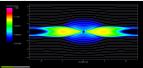
Adapted from "Annular FRC" PhD Proposal, C. (Niemela) Hill



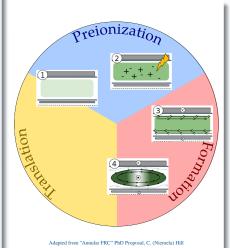


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- Field Reversal
 - Magnetic Reconnection



https://astrobear.pas.rochester.edu/trac/wiki/AstroBearProjects/resistiveMHD

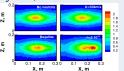


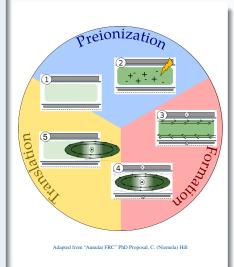




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- Plasmoid Ejection
 - $-\vec{j} \times \vec{B}$, Neutral Entrainment







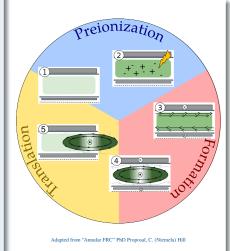


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Continuous Cycle: (5) impacts (1)







IMPORTANCE OF COLLISION PHYSICS



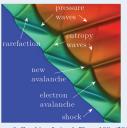
Important Collisions in Spacecraft Propulsion:

- Discharge and Breakdown in FRC
- Collisional Radiative Cooling/Ionization
- Combustion Chemistry

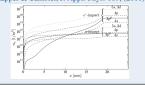
Common Features in Spacecraft Collisions:

- Relevant Densities Spanning Many Orders of Magnitude — 6+
- Transitions from Collisional to Collisionless
- Tiny Early e^- or Radical Populations Critical to Induction Delay
- Many types of Inelastic Collisions with Unknown Effects on Distribution Shapes

Shock Ionization



Kapper & Cambier, J. Appl. Phys. 109, (2011)





IMPORTANCE OF COLLISION PHYSICS



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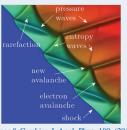
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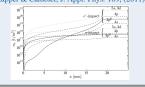
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Need Low Noise & High Dynamic Range Collision Algorithms

Shock Ionization



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STANDARD COLLISION MODELS



Previous Collision Methods:

- Monte Carlo Collisions (MCC)
 - Particles Collide with Background "Fluid"
 - Often Used in Plasma/PIC Simulation
 - Ion-e⁻ Collisions Assume Stationary Ions
 - No Conservation/Detailed Balance
- Direct Simulation Monte Carlo Collisions (DSMC)
 - Most Modern Versions use No-Time Counter (NTC) Method
 - Conservative/Reversible Collision
 - Satisfies Detailed Balance
 - Subset of Possible Collisions Sampled
 - Random Selection vs Z_{ij} for All/Nothing Collision

All Random Flip vs Number of Collisions: $Z_{ij} = \frac{n_i n_j}{2} \langle \sigma v \rangle dt$





Continuum to Discrete Representation:

ullet Many Particles $\widetilde{\rightarrow}$ Continuous Distribution







- ullet Many Particles $\widetilde{\rightarrow}$ Continuous Distribution
- Discretized VDF Yields Vlasov
 But Collision Integral Still a Problem







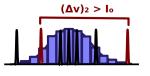
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Variable Weight "All-or-Nothing" Collisions?







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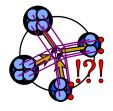
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Variable Weight "All-or-Nothing" Collisions?

Physically Inconsistent!

(Mixing Violates Momentum/Energy Conservation)





Stochastic Weighted Particle Method:

Developed by Rjasanow & Wagner

Attempted Collisions/Cell:

$$\nu = f(2\bar{w} - w_{min})N_p(N_p - 1) \langle \sigma v \rangle^{max} dt$$

Select Pair (i,j) if:

$$\begin{aligned} \text{Rand} &< \frac{w_i + w_j - w_{min}}{N_p(N_p - 1)(2\bar{w} - w_{min})} \\ &\quad \text{-or-} \\ \text{Rand} &< \frac{w_i + w_j - w_{min}}{(2w_{mor} - w_{min})} \end{aligned}$$

Collide If:

Rand
$$< \frac{\langle \sigma v \rangle_{ij}}{\langle \sigma v \rangle^{max}} \frac{f \max(w_i, w_j)}{w_i + w_j - w_{min}}$$

Perform Standard VHS Collisions

Generate/Modify Particles with:

$$\pm \Delta w/f = \pm \min(w_i, w_j)/f$$

Update $\langle \sigma v \rangle^{max}$





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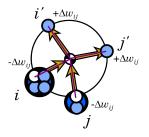


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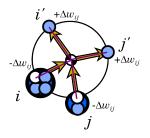


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- Still Requires Merge $w_i \neq \text{const}$

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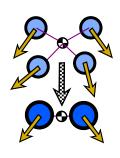


REVIEW OF CONSERVATIVE MERGE



Merge to Pair \rightarrow DOF for Conservation:

- (n+2):2 yields Exact Mass,
 Momentum, and Kinetic Energy
 Conservation
- Applied Spatially also Shown to Conserve Electrostatic Energy
- Though Energy Conserving, Still Thermalizes VDF



$$\begin{split} w_{cell} &= \sum_{i}^{(n+2)} w_{i} \\ \overline{\vec{v}} &= \frac{1}{w_{cell}} \sum_{i}^{(n+2)} w_{i} \vec{v}_{i} \\ \overline{V^{2}} &= \frac{1}{w_{cell}} \sum_{i}^{(n+2)} w_{i} \left(\vec{v}_{i} - \overline{\vec{v}} \right)^{2} \\ w_{(a/b)} &= w_{m}/2 \\ \vec{v}_{(a/b)} &= \overline{\vec{v}} \pm \hat{\mathcal{R}} \sqrt{\overline{V^{2}}} \\ \text{Similarly: } \vec{x}_{(a/b)} &= \overline{\vec{x}} \pm \hat{\mathcal{R}} \sqrt{\overline{x^{2}}} \end{split}$$



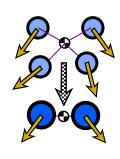
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Selection of Near Neighbors in VDF Limits Thermalization



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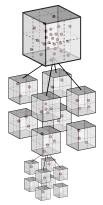
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Selection of Near Neighbors in VDF <u>Limits Thermalization</u>

Merge via Separate Octree/Species
Only Change for Mixtures!

Octree Velocity Bins



Efficient Neighbor Selection

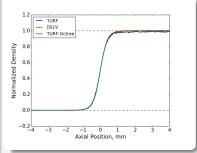


FROM RGD30: MACH 2 ARGON SHOCK



1D Normal Argon Shock Test

- Simple Verification vs. DS1V
- Initial Conditions: $T_0 = 293 \text{K}, n_0 = 1 \text{E} 22/\text{m}^3, v_0 = 637.4 \text{(m/s)}$
- Initial Jump to Post-Shock at 1cm
- VHS Collisions: T_{ref} =273K, d_{ref} =4.17Å, ω_{VHS} =0.81





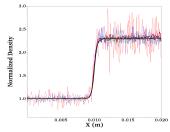
FROM RGD30: MACH 2 ARGON SHOCK



1D Normal Argon Shock Test

- Simple Verification vs. DS1V
- Initial Conditions: $T_0 = 293 \text{K}, n_0 = 1 \text{E} 22/\text{m}^3, v_0 = 637.4 \text{(m/s)}$
- Initial Jump to Post-Shock at 1cm
- VHS Collisions: T_{ref} =273K, d_{ref} =4.17Å, ω_{VHS} =0.81
- Time Average: \bar{n} from $t \in [80, 100) \mu s$

TURF - SWPM+Octree



Target N/Cell Quadrupled per Line



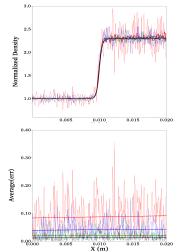
FROM RGD30: MACH 2 ARGON SHOCK



1D Normal Argon Shock Test

- Simple Verification vs. DS1V
- Initial Conditions: $T_0 = 293$ K, $n_0 = 1$ E22/m³, $v_0 = 637.4$ (m/s)
- Initial Jump to Post-Shock at 1cm
- VHS Collisions: T_{ref} =273K, d_{ref} =4.17Å, ω_{VHS} =0.81
- Time Average: \bar{n} from $t \in [80, 100)\mu s$
- Error (Normalized L₁): $err=|n-\bar{n}|/\bar{n}$
- Error Controlled: $err \propto \sqrt{N/cell}$

TURF - SWPM+Octree



Target N/Cell Quadrupled per Line

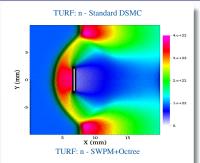


FROM RGD30: MACH 8 ARGON BOW SHOCK



2D Argon Shock Test

- Initial Conditions like M=2 Except: $v_0 = 2550 \text{m/s}$
- Specular: x=5-5.04mm with $y=\pm 2$ mm
- Half Domain Modeled: 80μm × 80μm Cells



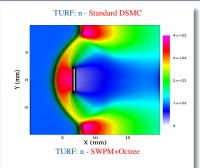


FROM RGD30: MACH 8 ARGON BOW SHOCK



2D Argon Shock Test

- Initial Conditions like M=2 Except: $v_0 = 2550 \text{m/s}$
- Specular: x=5-5.04mm with $y=\pm 2$ mm
- Half Domain Modeled: $80\mu m \times 80\mu m$ Cells
- Time Average: \bar{n} from $t \in [80, 100) \mu s$
- SWPM Similar to Standard DSMC



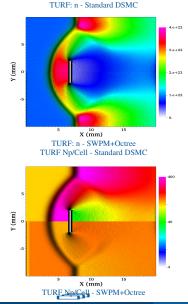


FROM RGD30: MACH 8 ARGON BOW SHOCK



2D Argon Shock Test

- Initial Conditions like M=2 Except: $v_0 = 2550 \text{m/s}$
- Specular: x=5-5.04mm with $y=\pm 2$ mm
- Half Domain Modeled: 80μm × 80μm Cells
- Time Average: \bar{n} from $t \in [80, 100)\mu s$
- SWPM Similar to Standard DSMC
- Despite Different Np/Cell

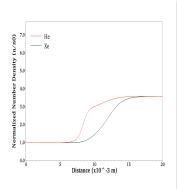






1D Normal He:Xe Shock Test

• Mach 3.89 with He:Xe of 97:3 (i.e. Bird '94 Fig 12.35)

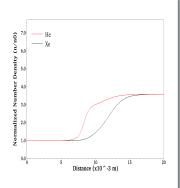


Converged 100x Baseline





- Mach 3.89 with He:Xe of 97:3 (i.e. Bird '94 Fig 12.35)
- Highlights Species Separation

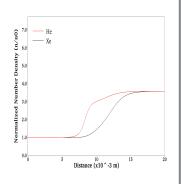


Converged 100x Baseline





- Mach 3.89 with He:Xe of 97:3 (i.e. Bird '94 Fig 12.35)
- Highlights Species Separation
- Separation Peak at $\rho_{He} \approx \rho_{Xe}$

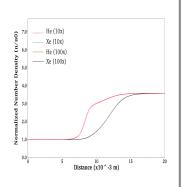


Converged 100x Baseline





- Mach 3.89 with He:Xe of 97:3 (i.e. Bird '94 Fig 12.35)
- Highlights Species Separation
- Separation Peak at $\rho_{He} \approx \rho_{Xe}$
- DSMC needs 33x He:Xe Macroparticles (From m_{Xe}/m_{He})
- Reduced Particle Count Introduces Error



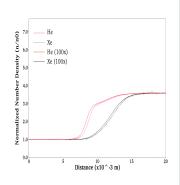
Comparison 10x vs. 100x





1D Normal He:Xe Shock Test

- Mach 3.89 with He:Xe of 97:3 (i.e. Bird '94 Fig 12.35)
- Highlights Species Separation
- Separation Peak at $\rho_{He} \approx \rho_{Xe}$
- DSMC needs 33x He:Xe Macroparticles (From m_{Xe}/m_{He})
- Reduced Particle Count Introduces Error
- Error in Time Average at Baseline

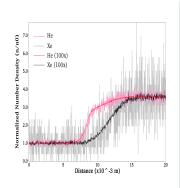


Comparison Baseline vs. 100x





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- Separation Peak at $\rho_{He} \approx \rho_{Xe}$
- DSMC needs 33x He:Xe Macroparticles (From m_{Xe}/m_{He})
- Reduced Particle Count Introduces Error
- Error in Time Average at Baseline
- Instantaneous: Dramatic Noise in Xe



Instantaneous Baseline vs. 100x

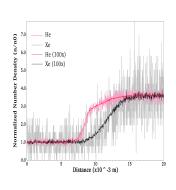




1D Normal He:Xe Shock Test

- Mach 3.89 with He:Xe of 97:3 (i.e. Bird '94 Fig 12.35)
- Highlights Species Separation
- Separation Peak at $\rho_{He} \approx \rho_{Xe}$
- DSMC needs 33x He:Xe Macroparticles (From m_{Xe}/m_{He})
- Reduced Particle Count Introduces Error
- Error in Time Average at Baseline
- Instantaneous: Dramatic Noise in Xe

Noise Reduction via Variable Weights?

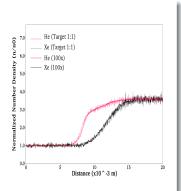


Instantaneous Baseline vs. 100x





- Xe Noise Controlled by 1:1 Target
- He:Xe Noise Comparable

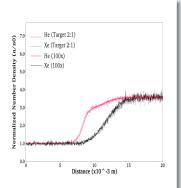


100x vs. FDSMC 1:1





- Xe Noise Controlled by 1:1 Target
- He:Xe Noise Comparable
- Direct Noise Control by Target Ratio

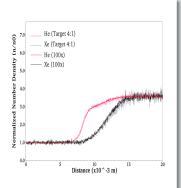


100x vs. FDSMC 2:1





- Xe Noise Controlled by 1:1 Target
- He:Xe Noise Comparable
- Direct Noise Control by Target Ratio

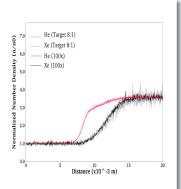


100x vs. FDSMC 4:1





- Xe Noise Controlled by 1:1 Target
- He:Xe Noise Comparable
- Direct Noise Control by Target Ratio

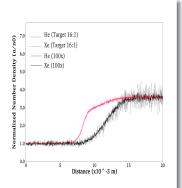


100x vs. FDSMC 8:1





- Xe Noise Controlled by 1:1 Target
- He:Xe Noise Comparable
- Direct Noise Control by Target Ratio



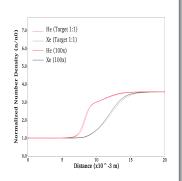
100x vs. FDSMC 16:1





1D He:Xe Shock with SWPM+Octrees

- Xe Noise Controlled by 1:1 Target
- He:Xe Noise Comparable
- Direct Noise Control by Target Ratio
- Converged Error Finite! (1:1 Target)
- Error Source Still Unidentified...



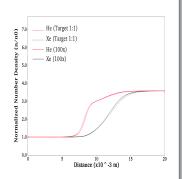
Averaged 100x vs. FDSMC 1:1





1D He:Xe Shock with SWPM+Octrees

- Xe Noise Controlled by 1:1 Target
- He:Xe Noise Comparable
- Direct Noise Control by Target Ratio
- Converged Error Finite! (1:1 Target)
- Error Source Still Unidentified...
- Potentially Sensitivity to He Tails?
 -Merge Impacts Higher Moments
 -TBD Error vs. He-Noise Level
 -Improvement Merge to Preserve Tails
- Adaptation of SWMP Incorrect?
 -# of Collisions Sampled as WDF Varies?
 -Collision Pair Rule Wrong, w_i ≪ w_i?



Averaged 100x vs. FDSMC 1:1

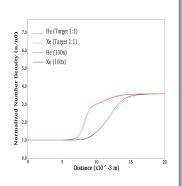




1D He:Xe Shock with SWPM+Octrees

- Xe Noise Controlled by 1:1 Target
- He:Xe Noise Comparable
- Direct Noise Control by Target Ratio
- Converged Error Finite! (1:1 Target)
- Error Source Still Unidentified...
- Potentially Sensitivity to He Tails?
 -Merge Impacts Higher Moments
 -TBD Error vs. He-Noise Level
 -Improvement Merge to Preserve Tails
- Adaptation of SWMP Incorrect?
 -# of Collisions Sampled as WDF Varies?
 -Collision Pair Rule Wrong, w_i ≪ w_i?

Error Identification in Future Work



Averaged 100x vs. FDSMC 1:1

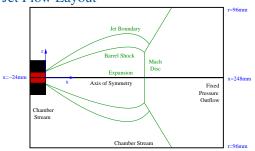


ROTHE'S HE: AR FREE-JET EXPERIMENT



- Helium-Argon Mixture
- Expanded through Nozzle to Vacuum
- e-Beam Concentration Measurements

Jet Flow Layout



Re=533

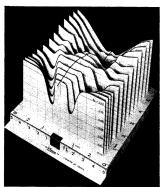


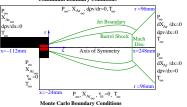
Fig. 11. Mole fraction of argon throughout the flow field of a free let.



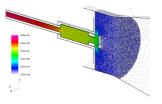
PRIOR RESULTS FROM DS2V AND CONTINUUM



Case Tested in PhD & RGD27:

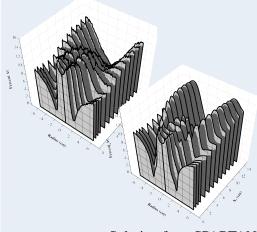


Surrogate for FRC Injection



Dynamic Range ≫ Shocks

Solution from DS2V



Solution from SPARTAN (Navier-Stokes + Diffusion Velocity)





Wing Increases near Nozzle Edge

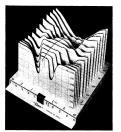
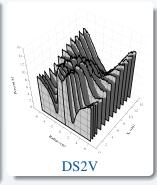
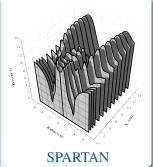


Fig. 11. Mole fraction of argon throughout the flow field of a free jet.

Experimental

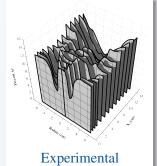


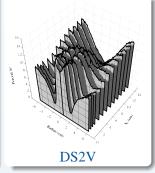


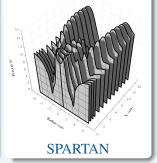




Wing Increases near Nozzle Edge



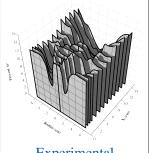




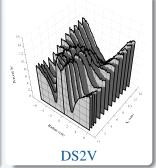


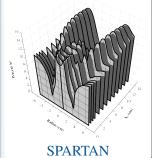


- Wing Increases near Nozzle Edge
- 2 Lower Radial Boundary Edge Concentration



Experimental

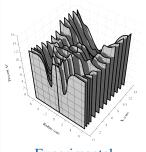




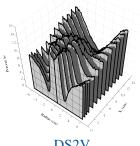




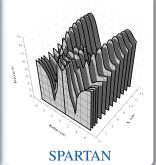
- Wing Increases near Nozzle Edge
- 2 Lower Radial Boundary Edge Concentration
- Deeper Jet Edge Concentration Drop



Experimental



DS2V







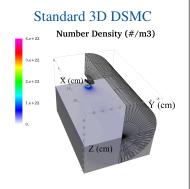
- TURF Naturally 3D Cartesian
- Considered 2D-Axisymmetric TURF
- Simple DSMC but Merge Complex (Conservation on $v_{\parallel}v_{\perp}$ -Quadtrees?)

Standard 3D DSMC





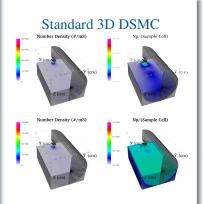
- TURF Naturally 3D Cartesian
- Considered 2D-Axisymmetric TURF
- Simple DSMC but Merge Complex (Conservation on $v_{\parallel}v_{\perp}$ -Quadtrees?)
- Opted to Run Coarse Full 3D (Simplified Boundary Conditions)
- 3D Expensive at Tractable Resolution
- Added Collision Sub-Cells to TURF







- TURF Naturally 3D Cartesian
- Considered 2D-Axisymmetric TURF
- Simple DSMC but Merge Complex (Conservation on $v_{\parallel}v_{\perp}$ -Quadtrees?)
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- 3D Expensive at Tractable Resolution
- Added Collision Sub-Cells to TURF
- Fractional DSMC Controls Np/Cell

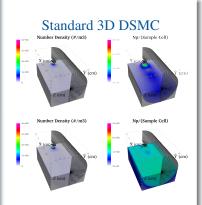


Multi-Species Fractional DSMC





- TURF Naturally 3D Cartesian
- Considered 2D-Axisymmetric TURF
- Simple DSMC but Merge Complex (Conservation on $v_{\parallel}v_{\perp}$ -Quadtrees?)
- Opted to Run Coarse Full 3D (Simplified Boundary Conditions)
- 3D Expensive at Tractable Resolution
- Added Collision Sub-Cells to TURF
- Fractional DSMC Controls Np/Cell
- Linear Density Obscures Results

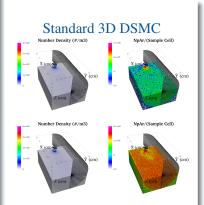


Multi-Species Fractional DSMC





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- Fractional DSMC Controls Np/Cell
- Linear Density Obscures Results
- Issue Clearer with NpAr/Cell



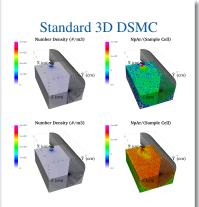
Multi-Species Fractional DSMC





- TURF Naturally 3D Cartesian
- Considered 2D-Axisymmetric TURF
- Simple DSMC but Merge Complex (Conservation on $v_{\parallel}v_{\perp}$ -Quadtrees?)
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- Fractional DSMC Controls Np/Cell
- Linear Density Obscures Results
- Issue Clearer with NpAr/Cell

 X_{Ar} to RX-Plane \rightarrow for Detailed Results...



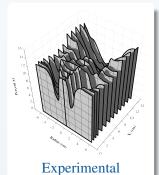
Multi-Species Fractional DSMC

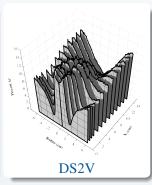


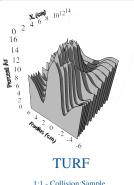
PRELIMINARY TURF RESULTS: HE:AR JET



Standard DSMC Poor Results







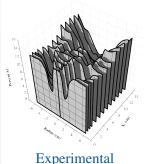
1:1 - Collision:Sample

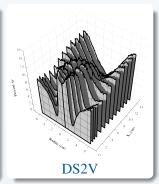


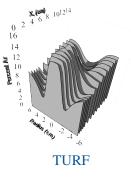
PRELIMINARY TURF RESULTS: HE:AR JET



- Standard DSMC Poor Results
- 2x2x2 Collision Cell Improves Standard DSMC







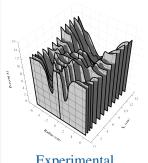
2x2x2:1 - Collision:Sample



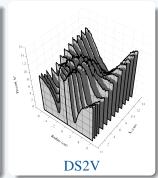
PRELIMINARY TURF RESULTS: HE:AR JET

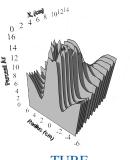


- Standard DSMC Poor Results
- 2 2x2x2 Collision Cell Improves Standard DSMC
- SWPM+Octree Significantly Better (2x2x2 Collision Cell)



Experimental





TURF

SWPM+Octree (2x2x2 Collision Cells)





The ∂f -Boltzmann for $f = f^{eq} + f^{dev}$

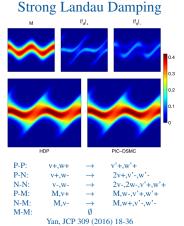
- ∂f Concept Old
- How to make ∂f cheaper than full-f(Must adapt DOF Usage..?)





The ∂f -Boltzmann for $f = f^{eq} + f^{dev}$

- ∂f Concept Old
- How to make ∂f cheaper than full-f(Must adapt DOF Usage..?)
- Recent Progress using $\pm \delta$ -weight Particles

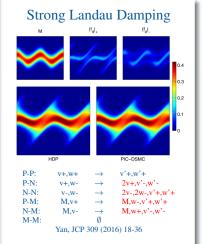






The ∂f -Boltzmann for $f = f^{eq} + f^{dev}$

- ∂f Concept Old
- How to make ∂f cheaper than full-f (Must adapt DOF Usage..?)
- Recent Progress using $\pm \delta$ -weight Particles
- Requires Remapping due to Particle Growth



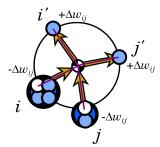




The ∂f -Boltzmann for $f = f^{eq} + f^{dev}$

- ∂f Concept Old
- How to make ∂f cheaper than full-f (Must adapt DOF Usage..?)
- Recent Progress using $\pm \delta$ -weight Particles
- Requires Remapping due to Particle Growth
- SWPM has Similar Issue (Basis for Octree B2B Collisions)

Stochastic Weight Particle Method (SWPM)



 $w_i = w_i - \Delta w_{ij} \& w_j = w_j - \Delta w_{ij}$ $w_{(N_p+1)} = \Delta w_{ij} \& w_{(N_p+2)} = \Delta w_{ij}$

+2 Particles/Collision

RGD30





The ∂f -Boltzmann for $f = f^{eq} + f^{dev}$

- ∂f Concept Old
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Octree ∂f -Boltzmann Bin-to-Bin Collisions



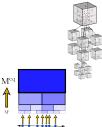




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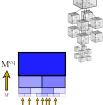




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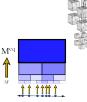




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Sample Collisions using Δw : for P-P, P-N, N-N, P-M, N-M M-M: \emptyset

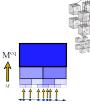




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- Collision Work $\propto \partial f$, not f

Octree ∂f -Boltzmann Bin-to-Bin Collisions





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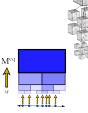
FUTURE DIRECTIONS: HYBRID ∂f



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- Valid at Adaptive Tree Depths
- Entropy Estimate for DOF Distribution

Octree ∂f -Boltzmann Bin-to-Bin Collisions





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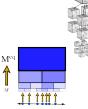




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Octree ∂f -Boltzmann Bin-to-Bin Collisions





Sample Collisions using Δw : for P-P, P-N, N-N, P-M, N-M M-M: \emptyset Combine with 2^6 -Tree

 $XV \rightarrow Multigrid Solves?$



Conclusion



Current Results:

- SWPM+Octree Option for Variable Weight Mixture Collisions
- Multiple Octree Merge only Modification for Multi-Species
- Initial Verification vs. Standard Shock Cases
- Merge/Target Enables Direct Control of Noise
- Unidentified Systematic Error with 1:1 Target
- Initial Testing on 3D Mixture Expansion Better with SWPM+Octree

Future Efforts:

- Additional Investigation of Error Source for Disparate Weights
- Improved Merge/Control of Tails
- Apply to Reacting Flow
- Adaptation for δf





Thank You

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Any opinions, finding, and conclusion or recommendations expressed in this material are those of the author and do not necessarily reflect the views of the United States Air Force.

Questions?